

**Table 13.1 Nozzle incremental drag (Ref. 10)**

Nozzle type	Subsonic $\frac{D/q}{A_{\text{fuselage}}^a}$
Convergent	0.036–0.042
Convergent iris	0.001–0.020
Ejector	0.025–0.035
Variable ejector	0.010–0.020
Translating plug	0.015–0.020
2-D nozzle	0.005–0.015

<sup>a</sup>Referenced to fuselage maximum cross-section area.